



SUPPLEMENTAL TYPE CERTIFICATE

10049666

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

FOKKER SERVICES B.V.

**HOEKSTEEN 40
2132 MS HOOFDORP
NETHERLANDS**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number : EASA.IM.A.120
Type Certificate Holder : THE BOEING COMPANY
Type Design - Model : 737-200, 737-200C, 737-300
737-400, 737-500, 737-600
737-700, 737-800
737-900, 737-900ER

Description of Design Change:

CPDLC (Controller Pilot Data Link Communication) in the Boeing 737

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Aviation Safety Agency,

Date of issue: 27 June 2014

Christian GUNITZBERGER
Deputy Section Manager /
Project Certification Manager
Large Aeroplanes

Note:

The following numbers are listed on the certificate:
EASA current Project Number: 0010028488-001

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Associated Technical Documentation:

Certification Plans: CP-0212 Issue 3, CP-213 Issue 1, CP-0219 Issue 2
Compliance Record Reports: CRR-0212 Issue 1, CRR-0213 Issue 1, CRR-0219 Issue 1
SE-779 Issue 1: Boeing 737 Classic - GPS antenna ICA Supplement
733-EXS-AFM-S-001 Revision 1: AFM Supplement 737 Classic
738-EXS-AFM-S-001 Revision 1: AFM Supplement 737 NG
or later revisions of the above listed documents approved by EASA.

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- end -

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