



## SUPPLEMENTAL TYPE CERTIFICATE

**EASA.A.S.03317**

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EC) No. 1702/2003 to

**Fokker Services B.V.**

**P.O. Box 231  
2150 AE Nieuw Vennepe  
The Netherlands**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

**Original Product Type Certificate Number:** EASA.A.064

**Type Certificate Holder:** Airbus

**Model:** A318-112

**Description of Design Change:**

A318-112 VIP conversion, for description see Chapter 1.1 of. Compliance Record Report A18-MAK-CRR-0001 issue 5 d.d. 29-5-2009.

**Associated Technical Documentation:**

- Compliance Record Report A18-MAK-CRR-0001 issue 5 d.d. 29-5-2009
- AFM Supplement Airbus A318 Model A318-112 Aircraft for the VIP Conversion As installed in Serial number 3238 issue 5 d.d. 29-5-2009
- MMEL Supplement Airbus A318 Model A318-112 Aircraft for the VIP Conversion As installed in Serial number 3238 issue 4 d.d. 29-5-2009
- Instructions for Continued Airworthiness SE-702 issue 1 d.d. 19-5-2009

**Limitations and Conditions:**

1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is subject to written permission of the approval holder.
2. Refer to Chapter 2, Limitations of the AFM Supplement Airbus A318 Model A318-112 Aircraft for the VIP Conversion As installed in Serial number 3238 issue 5 d.d. 29-5-2009
3. The STC is approved only for the product configuration as defined in the approved design data referred to in paragraph "Description" Compatibility with other aircraft / engine configuration shall be determined by the installer.

This Certificate shall remain valid unless otherwise surrendered or revoked.

**For the European Aviation Safety Agency,**

**Date of issue: 29 May 2009**

**Christian GUNITZBERGER  
Project Certification Manager  
Large Aeroplanes**

STC – EASA.A.S.03317 – Fokker Services B.V.